

Human Missions to Mars

Enabling Technologies for Exploring the Red Planet

Donald Rapp

Human Missions to Mars

Enabling Technologies for Exploring the Red Planet



Published in association with
Praxis Publishing
Chichester, UK



Dr Donald Rapp
Independent Contractor
South Pasadena
California
USA

SPRINGER-PRAXIS BOOKS IN ASTRONAUTICAL ENGINEERING
SUBJECT *ADVISORY EDITOR*: John Mason, M.Sc., B.Sc., Ph.D.

ISBN 978-3-540-72938-9 Springer Berlin Heidelberg New York

Springer is part of Springer-Science + Business Media (springer.com)

Library of Congress Control Number: 2007932001

Apart from any fair dealing for the purposes of research or private study, or criticism or review, as permitted under the Copyright, Designs and Patents Act 1988, this publication may only be reproduced, stored or transmitted, in any form or by any means, with the prior permission in writing of the publishers, or in the case of reprographic reproduction in accordance with the terms of licences issued by the Copyright Licensing Agency. Enquiries concerning reproduction outside those terms should be sent to the publishers.

© Praxis Publishing Ltd, Chichester, UK, 2008
Printed in Germany

The use of general descriptive names, registered names, trademarks, etc. in this publication does not imply, even in the absence of a specific statement, that such names are exempt from the relevant protective laws and regulations and therefore free for general use.

Cover design: Jim Wilkie
Project management: Originator Publishing Services Ltd, Gt Yarmouth, Norfolk, UK

Printed on acid-free paper

Contents

Preface	xv
List of figures	xvii
List of tables	xxiii
List of abbreviations and acronyms	xxvii
1 Why explore Mars?	1
1.1 Robotic exploration—the Establishment’s view	1
1.2 The curmudgeon’s view on the search for life on Mars	5
1.3 Why send humans to Mars?—the enthusiast’s view	7
1.4 Sending humans to Mars—the skeptic’s view	10
2 Planning space campaigns and missions.	13
2.1 Campaigns	13
2.2 Planning space missions	14
2.3 Architectures	15
2.4 A mission as a sequence of steps	19
2.5 What’s at the destination?	22
2.6 What’s in low Earth orbit	24
2.7 What’s on the launch pad?	25
2.8 IMLEO requirements for space missions	25
3 Getting there and back	27
3.1 Propulsion systems	27
3.1.1 Propellant requirements for space transits	28
3.1.2 The rocket equation	30
3.1.3 Dry mass of rockets	33

3.2	Trajectory analysis	35
3.2.1	Rocket science 101	35
3.2.1.1	Constants of motion	35
3.2.1.2	Energy of an orbit	36
3.2.1.3	Measured values for the Sun, Earth, and Mars	38
3.2.1.4	Escaping the influence of a planet	39
3.2.1.5	Earth and Mars solar orbit velocities	40
3.2.1.6	Hohmann transfer to Mars	40
3.2.1.7	Earth and Mars low-orbit velocities	42
3.2.1.8	Earth escape	42
3.2.1.9	Mars orbit insertion—part 1	43
3.2.1.10	Summary of transfers to and from Hohmann orbit	44
3.2.1.11	Orbital period	45
3.2.1.12	Mars orbit insertion—part 2	47
3.2.1.13	Summary.	49
3.2.2	Mars mission duration and propulsion requirements	50
3.2.3	More realistic models	53
3.3	Earth to low Earth orbit	54
3.4	Departing from LEO	58
3.4.1	The Δv requirement	58
3.4.2	Mass sent toward Mars	59
3.4.3	Nuclear thermal rocket for TMI	61
3.4.4	Solar electric propulsion for orbit raising	63
3.5	Mars orbit insertion	65
3.6	Ascent from the Mars surface	68
3.7	Trans-Earth injection from Mars orbit	69
3.8	Earth orbit insertion	69
3.9	Gear ratios	71
3.9.1	Introduction	71
3.9.2	Gear ratio calculations	71
3.9.3	Gear ratio for Earth departure	73
3.10	LEO to Mars orbit	73
3.11	LEO to the Mars surface	76
3.12	IMLEO for Mars missions	76
3.12.1	Chemical propulsion and aero-assist	76
3.12.2	Use of Nuclear Thermal Propulsion	79
3.12.3	Use of ISRU	82
4	Critical Mars mission elements	85
4.1	Life support consumables	85
4.1.1	Consumable requirements (without recycling)	85
4.1.2	Use of recycling systems	87
4.1.3	Life support summary	90

4.2	Radiation effects and shielding requirements	91
4.2.1	Radiation sources	91
4.2.2	Definitions and units	92
4.2.3	Radiation effects on humans and allowable dose	93
	4.2.3.1 Allowable exposure	93
	4.2.3.2 Radiation effects on humans	94
4.2.4	Confidence intervals and point estimates	95
4.2.5	Radiation in space	96
4.2.6	Radiation levels in Mars missions	97
4.2.7	Radiation summary	99
4.3	Effects of microgravity	99
4.3.1	The Whedon–Rambaut review of low- <i>g</i> effects	99
4.3.2	Artificial gravity	101
4.3.3	NASA plans for low- <i>g</i> effects	105
4.4	Abort options and mission safety	106
4.4.1	Abort options and mission safety in ESAS lunar missions	106
4.4.2	The Mars Society Mission	108
4.4.3	Abort options conclusions	109
4.5	Habitats	111
4.5.1	DRM-1 Habitats	111
4.5.2	DRM-3 Habitats	112
4.5.3	Dual Landers Habitat	113
4.5.4	The “MOB” design	113
	4.5.4.1 Basic design	113
	4.5.4.2 ISRU	117
	4.5.4.3 Habitat mass	118
	4.5.4.4 Habitat structure	119
	4.5.4.5 ECLSS	120
4.5.5	Sasakawa International Center for Space Architecture	121
	4.5.5.1 Habitat configurations	121
	4.5.5.2 Logistic and crew support	125
4.5.6	Inflatable Habitats	125
4.6	Aero-assisted orbit insertion and entry, descent, and landing	127
4.6.1	Introduction	127
4.6.2	Challenges for aero-assist technology	131
	4.6.2.1 Atmospheric variations	132
	4.6.2.2 Aeroshell shape and size	132
	4.6.2.3 Supersonic parachute	134
	4.6.2.4 Landing accuracy	135
4.6.3	Entry, descent, and landing requirements for human missions to Mars	135
	4.6.3.1 Initial results	135
	4.6.3.2 Updated results: 2006	140
4.6.4	Precision landing	143
4.6.5	Development, test, and validation program	145

5	<i>In situ</i> utilization of indigenous resources	149
5.1	Value of ISRU	149
5.2	Lunar ISRU	150
5.2.1	Introduction	150
5.2.2	Potential products of ISRU	151
5.2.2.1	Ascent propellants	151
5.2.2.2	Life support consumables	152
5.2.2.3	Propellants delivered to LEO from the Moon	153
5.2.2.4	Propellants delivered to lunar orbit for descent (and ascent)	154
5.2.2.5	Regolith for radiation shielding	155
5.2.2.6	Visionary concepts	155
5.2.3	Lunar resources	158
5.2.4	Lunar ISRU processing	159
5.2.4.1	Oxygen from FeO in regolith	159
5.2.4.2	Oxygen from regolith silicates	160
5.2.4.3	Extracting putative volatiles	161
5.2.4.4	Utilizing polar ice deposits	162
5.2.5	Cost analysis for ESAS lunar ISRU	168
5.2.6	A new paradigm for lunar ISRU	170
5.3	Mars ISRU	172
5.3.1	Introduction	172
5.3.2	Timeline for ISRU on Mars	173
5.3.3	Mars ISRU products	175
5.3.4	Mars ISRU processes	180
5.3.4.1	Oxygen-only processes	180
5.3.4.2	The Sabatier/Electrolysis process	185
5.3.4.3	Timeline for ISRU	187
5.3.4.4	CO ₂ acquisition from the Mars atmosphere	187
5.3.5	Mass and power requirements of a Mars ISRU system	188
5.4	Fueling Mars-bound vehicles from lunar resources	190
5.4.1	Introduction	190
5.4.2	Value of lunar water in LEO	190
5.4.3	Percentage of water mined on the Moon transferred to LEO	192
5.4.3.1	Transfer via LL1	192
5.4.3.2	Dependence on junction site	196
5.5	Lunar ferry for descent propellants	200
5.6	Staging, assembly, and refueling in near-Earth space	202
5.6.1	Introduction	202
5.6.2	Michael Griffin and orbiting fuel depots	203
5.6.3	Propellants for Earth departure	204
5.6.4	On-orbit staging	206
5.7	Transporting hydrogen to Mars	209
5.7.1	Terrestrial vs. space applications	209

5.7.2	Storage of hydrogen in various physical and chemical states	211
5.7.2.1	Storage as high-pressure gas at about room temperature	211
5.7.2.2	Storage as a cryogenic liquid	213
5.7.2.3	Storage as a dense gas at reduced temperature	215
5.7.2.4	Storage as solid hydrogen	216
5.7.2.5	Storage as solid–liquid slush	217
5.7.2.6	Storage as hydrogen at its triple point	218
5.7.2.7	Storage as adsorbed hydrogen on a sorbent	218
5.7.2.8	Storage in metal hydrides	220
5.7.2.9	Storage in glass microspheres	221
5.7.3	Boil-off in space	222
5.7.3.1	Rate of boil-off from MLI tanks	222
5.7.3.2	Mass effect of boil-off in space	225
5.7.3.3	Boil-off rate on Mars	227
5.7.4	Zero boil-off systems	228
5.7.5	Potential space applications for hydrogen	231
5.7.5.1	Shuttle H ₂ tanks	231
5.7.5.2	Hydrogen storage trade study for a solar electric orbital transfer vehicle	233
5.7.5.3	Estimates of boil-off by a lunar mission focused study	233
5.7.5.4	Transporting hydrogen to the lunar surface	234
5.7.5.5	Transporting hydrogen to Mars for ISRU	234
5.7.6	Summary and conclusions	235
6	Mars mission analysis	237
6.1	Previous Mars mission studies	237
6.1.1	Office of Exploration case studies (1988)	237
6.1.2	Office of Exploration case studies (1989)	238
6.1.3	NASA 90-Day Study (1989)	239
6.1.4	America at the threshold—“The Synthesis Group” (1991)	239
6.1.5	First lunar outpost (1993)	240
6.1.6	Human lunar return (1996)	241
6.1.7	Mars Exploration Design Reference Missions (1994–1999)	241
6.1.8	Decadal Planning Team/NASA Exploration Team (2000–2002)	242
6.1.9	Integrated Space Plan (2002–2003)	242
6.1.10	Exploration Systems Mission Directorate (2004–2005)	242
6.1.11	Constellation (2006–?)	246
6.2	Mars Design Reference Missions (DRMs)	248
6.2.1	DRM-1	248
6.2.1.1	Introduction	248
6.2.1.2	DRM-1 vehicles	249

6.2.1.3	Mission sequence	250
6.2.1.4	Mars Ascent Vehicle (MAV)	252
6.2.1.5	Surface power	253
6.2.1.6	Interplanetary transportation system	253
6.2.1.7	Launch vehicles	255
6.2.1.8	<i>In situ</i> resource utilization	256
6.2.1.9	Life support consumables	258
6.2.2	DRM-3	258
6.2.3	Mass comparisons: DRM-3 and DRM-1	261
6.2.4	The Dual Landers Reference Mission	262
6.2.5	Common characteristics of 1990s' JSC Mars DRMs	266
6.3	Mars DRMs in the 2005–2006 era	268
6.3.1	The ESAS Report Mars plans	268
6.3.2	2006 Mars DRM	272
6.4	Non-NASA Design Reference Missions	274
6.4.1	Mars Direct	274
6.4.2	The Mars Society Mission	278
6.4.3	TeamVision approach to space exploration	281
6.4.4	The MIT Study	282
6.4.5	ESA Concurrent Design Facility Study (2003)	284
6.5	IMLEO for human missions to Mars	286
7	How NASA is dealing with return to the Moon	291
7.1	Introduction	291
7.1.1	Need for a Heavy Lift Launch Vehicle	291
7.1.2	Urgency in returning to the Moon at the earliest possible opportunity	293
7.2	Moon–Mars connection	293
7.2.1	The Moon as a means of risk reduction for Mars	293
7.2.2	ISRU as a stepping-stone from the Moon to Mars	295
7.2.3	The ESAS viewpoint on the Moon–Mars connection	298
7.2.4	A European viewpoint	299
7.3	Lunar architecture options	300
7.3.1	Zubrin's analysis.	300
7.3.2	MIT analysis	303
7.3.3	The Broad Study	308
7.3.4	The Focused Study	311
7.3.5	Comparison of Broad and Focused Study estimates of Δv	314
7.3.6	ESAS Report Δv estimates	314
7.3.7	Lunar orbit rendezvous vs. L1 rendezvous or direct return	319
	7.3.7.1 Broad Study	319
	7.3.7.2 Focused Study	319
7.3.8	ESAS report architecture comparison	320
7.4	The ESAS lunar architecture	324
7.4.1	Description	324

7.4.2	Independent mass estimates	324
7.5	Do we need a CEV?	326
7.6	The NASA lunar architecture process	329
8	Why the NASA approach will likely fail to send humans to Mars prior to c. 2080	331
8.1	Major differences between Mars and Lunar missions	331
8.2	Is NASA too busy for Mars missions?	332
8.3	Mars architecture planning	334
8.4	Need for new technology	340
8.4.1	Space Science Enterprise (SSE)	340
8.4.1.1	SSE scope of technology	340
8.4.1.2	Lead Centers	344
8.4.1.3	On again/off again	346
8.4.1.4	SSE technology summary	347
8.4.2	Human exploration technology	348
8.4.3	Exploration Systems Research and Technology (ESRT)	349
8.5	2020 vision	351
8.6	Human exploration strategy.	354
8.6.1	Background	354
8.6.1.1	Destination-driven vs. constituency-driven programs	354
8.6.1.2	The 2004–2005 Human Exploration Initiative	355
8.6.2	Current NASA plans (2006–2007)	357
8.6.3	Recommended strategy	358
8.6.3.1	Propulsion systems	359
8.6.3.2	Aero-assisted entry, descent, and landing	360
8.6.3.3	Habitats and capsules	361
8.6.3.4	Life support consumables	361
8.6.3.5	<i>In situ</i> resource utilization	362
8.6.3.6	Power systems	363
8.6.3.7	Site selection	363
8.6.3.8	Precursor missions at Mars	364
8.6.3.9	Mitigation of dust effects	366
8.6.3.10	Use of Moon for testing and demonstrations	366
8.6.3.11	Testing in Earth orbit	367
8.7	Conclusions	368

APPENDICES

A	Solar energy on the Moon	369
A.1	First approximation to lunar orientation	369
A.2	Solar insolation on a horizontal surface	371
A.3	Solar insolation on a vertical surface	373
A.4	Insolation on a surface tilted at latitude angle toward the equator	373

A.5	A surface that is always perpendicular to the solar rays	374
A.6	Effect of non-ideal lunar orbit	377
A.7	Operating temperature of solar arrays on the Moon	378
A.8	Solar energy systems at the equator	378
A.8.1	Short-term systems (<354 hours)	378
A.8.2	Long-term systems (>354 hours)	380
A.9	Effects of dust	382
A.10	Solar energy systems in polar areas	383
A.10.1	Polar sites	383
A.10.2	GRC solar polar study	386
B	Solar energy on Mars	389
B.1	Solar intensities in current Mars orbit	389
B.1.1	Introduction	389
B.1.2	Irradiance in a clear atmosphere	391
B.1.3	Effect of atmosphere	391
B.1.3.1	The direct beam	393
B.1.3.2	Simple “two-flux” model of scattering and absorption of sunlight in the Mars atmosphere	395
B.1.3.3	Sophisticated model of scattering and absorption of sunlight in the Mars atmosphere	398
B.2	Solar intensities on a horizontal and tilted surfaces.	400
B.2.1	Nomenclature	402
B.2.2	Solar intensity on a horizontal surface	403
B.3	Solar intensities on a fixed tilted surface	404
B.3.1	The diffuse component on a tilted surface	405
B.3.2	Reflection from ground in front of tilted collector	407
B.3.3	Total intensity on a tilted surface	407
B.3.4	Rotating tilted surfaces	407
B.4	Numerical estimates of solar intensities on Mars	408
B.4.1	Solar energy on horizontal surfaces	408
B.4.1.1	Daily total insolation	408
B.4.1.2	Hourly insolation patterns on a horizontal surface	411
B.4.1.3	Total insolation on a horizontal surface over a Martian year	411
B.4.2	Solar intensities on sloped surfaces	411
B.4.2.1	Fixed slope surfaces	411
B.4.2.2	Rotating Sun-facing tilted planes	415
B.4.3	Solar energy on Mars over the last million years	418
B.4.3.1	Variations in the Mars orbit	418
B.4.3.2	Insolation on horizontal surfaces over a million years	419
B.4.3.3	Insolation on tilted surfaces over a million years	421

B.5	Effect of dust on array surfaces—simple models	425
B.5.1	Introduction	425
B.5.2	Optical depth	426
B.5.3	Particle size distribution	426
B.5.4	Number of dust particles in vertical column	428
B.5.5	Rate of settling of dust particles	428
B.5.6	Initial rate of obscuration	429
B.5.7	Longer term buildup of dust	429
B.6	Pathfinder and MER data on dust obscuration	432
B.7	Aeolian removal of dust from surfaces	438
B.8	Obscuration produced by dust on solar arrays	440
B.8.1	JPL experiments—2001	440
B.8.2	Summary and conclusions on dust obscuration	441
C	Water on Mars	445
C.1	Introduction	445
C.2	Background information	448
C.2.1	Temperatures on Mars	448
C.2.2	Pressures on Mars	451
C.2.3	Water vapor concentrations on Mars	453
C.3	Equilibrium models for subsurface ice	455
C.3.1	Introduction	455
C.3.2	Models for stability of subsurface ground ice on Mars— current conditions	457
C.3.3	Long-term evolution of water on Mars	464
C.3.4	Effect of Mars orbit variations during the past ~1 million years	465
C.3.5	Evolution of south polar cap	471
C.4	Experimental detection of water in Mars' subsurface by neutron spectroscopy from orbit	472
C.4.1	Introduction	472
C.4.2	Original data reduction	474
C.4.3	Modified data reduction	474
C.4.4	Water content based on uniform regolith model	475
C.4.5	Water content based on a two-layer regolith model at equatorial and mid-latitudes	475
C.4.6	Interpretation of depth from neutron measurements	476
C.5	Comparison of neutron data with physical properties of Mars	478
C.5.1	Surface and atmospheric properties	478
C.5.2	Water deposits vs. topography at low and mid-latitudes	481
C.5.3	Seasonal distribution of equatorial near-surface water	484
C.6	The polar caps	484
C.7	Liquid water on Mars	486
C.7.1	Regions where surface temperature excursions exceed 273.2 K	486

C.7.2	Liquid water below the surface	491
C.7.3	Imaging indications of recent surface water flows	493
C.7.3.1	Gullies	493
C.7.3.2	Surface streaks	496
C.8	Evidence from craters	498
C.8.1	Introduction	498
C.8.2	The work of Nadine Barlow (and friends)	500
C.8.3	Other crater studies	505
C.8.4	Commentary	506
C.9	Summary	510
Index	513

Preface

The basis of science is observation. By making repeated observations of relevant phenomena, we discover patterns in the behavior of natural systems, and from this we formulate theories that allow us to predict the outcome of future observations.

While this description of science works well for topics that have been heavily explored and are fairly well understood, there are many occasions in the real world at the cutting edge of knowledge where we have not yet made enough observations to draw conclusions, or where the phenomena of interest occurred so long ago that it is extremely difficult to fathom what might have actually taken place in ancient epochs.

One important case in point is the question of how, where, and when life originated from inanimate matter. The one basic piece of data that we have is that we know that life existed in a primitive form on Earth over 3 billion years ago (BYA). This was determined from fossil remains of early forms of life in dated deposits. Some of the questions that arise include:

- Where and when did life on Earth begin (on Earth, or somewhere else)?
- What was the process by which inanimate matter formed life in its primitive form?
- Is there life elsewhere in the solar system or external to the solar system?

All of these questions are in some sense subservient to the big question:

- Is the formation of life from inanimate matter a likely (or even a deterministic) process, given a reasonable length of time and a warm climate, liquid water, and a smattering of chemical elements in the lower periodic table?

In addressing these questions, some scientists have applied analysis and imagination to conceive of a wide variety of hypothetical scenarios for the formation of life, none of which is supported by evidence, and all of which seem highly dubious in retrospect. The world of science does not seem to be able to flatly say: “We simply do not

understand much about the formation of life,” and leave it at that. Just as nature abhors a vacuum, so does science abhor the absence of answers to vital questions. As a result, the science community has jumped on a series of bandwagons over the past several decades purporting to “explain” how life began, although many of these have now fallen by the wayside. Nevertheless, it is widely believed that formation of life from inanimate matter is a likely (or even a deterministic) process, and, given that belief, the most natural place to seek extraterrestrial life is Mars. Thus, the NASA exploration programs are heavily centered on a search for life on Mars.

Human missions to Mars represent the pinnacle of solar system exploration for the next half-century. In addition to providing a means of searching for life on Mars, such missions would represent an inspiring engineering achievement, and create a new era of expansion of humanity into space. Because such missions would require a major technological effort as well as very large expenditures, they remain for the moment as futuristic concepts embodied in paper studies by advocates and enthusiasts.

In the world of science and engineering, there is room for advocates and skeptics. Advocates play an important role in imagining what might be, and stubbornly pursue a dream that may be difficult to realize, but which in the end may be achievable. Skeptics identify the barriers, difficulties, pitfalls, and unknowns that impede the path, and point out the technical developments needed to enable fulfillment of the dream.

In the realm of human missions to Mars, we have a number of studies by advocates and enthusiasts, but there seems to be a total absence of skepticism in this arena. This book represents the first skeptical analysis of human missions to Mars, and it is offered as a counter-balance to the optimism so widely promulgated by the NASA Johnson Space Center, the Mars Society, and others.

Donald Rapp
April 2007

Figures

2.1	NASA ESAS architecture for lunar sortie missions	16
2.2	ESAS lunar sortie architecture	17
2.3	MIT lunar architectures.	18
2.4	States and steps for Mars mission.	21
2.5	Typical Δv for lunar sortie missions	22
2.6	Typical Δv (km/s) for Mars missions.	23
2.7	Vehicles in JSC's Mars DRM-3	23
3.1	Model of a rocket burn	28
3.2	Specific impulse of methane/oxygen rocket	color
3.3	Flight path at periapsis showing flight path is perpendicular to radius	36
3.4	Spacecraft in orbit about a planet.	36
3.5	Force balance for a circular orbit	37
3.6	Typical hyperbolic trajectory	38
3.7	Escape from Earth	39
3.8	Transfer from moving with the Earth to Hohmann orbit	41
3.9	Transfer from Earth orbit to Hohmann orbit.	45
3.10	Kepler's second law.	46
3.11	Properties of the ellipse	47
3.12	Mars orbit insertion	49
3.13	Hyperbolic and elliptical trajectories at Mars.	50
3.14	Typical "pork-chop" plot for 2022	color
4.1	Probability of radiation exposure induced death.	96
4.2	Dose for GCR at Solar Minimum	96
4.3	BFO dose equivalent vs. lunar regolith thickness	97
4.4	Parameters for artificial gravity.	102
4.5	Concept for artificial gravity.	103
4.6	Dual connected Habitats in DRM-1	113
4.7	DRM-3 Habitat	114
4.8	Dual Landers vehicle.	116
4.9	MOB concept for Habitat	117

4.10	MOB plan for upper floor	118
4.11	MOB plan for lower floor	118
4.12	SICSA layout/configuration options	123
4.13	SICSA logistic module concept	123
4.14	Habitat module geometry	124
4.15	Inflatable Habitat concept	126
4.16	MGS aerobraking process	129
4.17	Sequence of events in aerocapture	130
4.18	Typical aeroshell	133
4.19	Mass of entry system for Mars	139
4.20	Elements of Mars entry system	139
4.21	Payload mass fraction to the Martian surface	142
5.1	Time-line for a Mars mission	174
5.2	Zirconia cell	181
5.3	Ion current from zirconia cell	182
5.4	Composition vs. temperature in RWGS process	184
5.5	Composition vs. temperature in Sabatier process	185
5.6	Earth–Moon Lagrange Points	191
5.7	Transporting water from the Moon to LEO	191
5.8	Potential saving in IMLEO	202
5.9	Density of gaseous hydrogen	215
5.10	Isotherms for hydrogen adsorption	219
5.11	Boil-off rate for hydrogen	224
5.12	Performance of insulation vs. pressure	227
5.13	Tank insulation	230
6.1	Mission sequence for Mars DRM-1	252
6.2	Solar electric propulsion orbit-raising with a solar electric propulsion tug	265
6.3	Dual Landers mission	267
6.4	ESAS Version of Mars DRM	269
6.5	MIT architectures for Mars missions	284
7.1	Lunar sortie missions	302
7.2	Lunar architectures in the MIT Study	304
7.3	Lunar orbit insertion	309
7.4	Δv for lunar orbit insertion—no loitering	316
7.5	Δv for lunar orbit insertion—loitering	317
7.6	Plane change for worst-case rendezvous	318
7.7	ESAS Report estimates of probability of loss of crew	color
7.8	ESAS Report estimates of probability of loss of mission	color
7.9	1.5-launch EOR–LOR architecture	325
8.1	Campaign for human exploration of Mars	335
8.2	NASA Mars DRM (c. 2006)	color
8.3	2006 NASA Mars DRM	color
A.1	The Moon in the ecliptic plane	370
A.2	Motion of Earth and Moon	370
A.3	Lunar orbit in ecliptic plane	371
A.4	Horizontal surface on the Moon	372
A.5	Solar intensity on a horizontal surface	372
A.6	Solar intensity on a horizontal surface—contour plot	373
A.7	Total energy per 14.77-day cycle	374

A.8	Vertical surface on the Moon	374
A.9	Tilted surface on the Moon	375
A.10	Tilted plane on the Moon	375
A.11	Surface perpendicular to solar rays	376
A.12	Rotating plane that always faces the Sun.	376
A.13	Solar intensity for rotating planes.	377
A.14	Tent array structure.	379
A.15	Solar power generated by tent array	379
A.16	Effect of high nighttime power fraction	381
A.17	Effect of low nighttime power fraction	381
A.18	Solar intensity through a dust layer	383
A.19	Deposition of dust near Apollo	383
A.20	Elevation of the Sun at the lunar pole.	384
A.21	The height needed to see over the pole in winter	384
A.22	Illumination map of the lunar south pole	color
A.23	Craters that contain permanent shadow	color
A.24	Illumination map of the Moon's north pole.	color
B.1	Mars axis of rotation.	390
B.2	An element of path length	393
B.3	Optical depth	394
B.4	Simple model of the Mars atmosphere	396
B.5	Transmission coefficients of the two-flux model	399
B.6	Mars atmospheric transmission.	401
B.7	Comparison of atmosphere transmission models	401
B.8	Tilted array surface on Mars	404
B.9	Photograph of Sun near sunset by Mars Pathfinder	color
B.10	Photograph of Sun near sunset by Mars Pathfinder	color
B.11	Parameters for tilted solar array.	406
B.12	Fraction of sky "seen" by tilted array.	406
B.13	Reflected solar rays on tilted array	407
B.14A	Insolation for optical depth = 0.3	408
B.14B	Insolation for optical depth = 0.5	409
B.14C	Insolation for optical depth = 1.0	409
B.14D	Insolation for optical depth = 2.0	410
B.14E	Insolation for optical depth = 4.0	410
B.15A	Hourly solar intensities at 70°N	412
B.15B	Hourly solar intensities at 70°S.	412
B.15C	Hourly solar intensities at 15°N	413
B.15D	Hourly solar intensities at 15°S.	413
B.15E	Hourly solar intensities at 45°N	414
B.15F	Hourly solar intensities at 45°S.	414
B.16	Yearly insolation for various optical depths.	415
B.17	Solar intensity on an equator-facing slope	416
B.18	Solar intensity on a pole-facing slope	416
B.19	Pole-facing and equator-facing slopes	417
B.20	Solar intensity on tilted surface at 80°S	417
B.21	Solar intensity on tilted surface at 65°N	418
B.22	Historical variation of Mars orbit	419
B.23	Historical insolation in the northern hemisphere	420

B.24	Historical insolation in the southern hemisphere	420
B.25	Comparison of southern and northern hemispheres	421
B.26	Historical insolation tilt toward equator—south.	422
B.27	Historical insolation tilt toward equator—north.	423
B.28	Historical insolation tilt toward pole—north	423
B.29	Historical insolation tilt toward pole—south	424
B.30	Optical depths on Pathfinder mission	426
B.31	Optical depths on MER mission.	427
B.32	Optical depth on Viking Landers	427
B.33	Distribution of dust particles by layers	430
B.34	Obscuration vs. sol for multiple layers.	431
B.35	Dependence of obscuration on dust loading.	431
B.36	Obscuration on Pathfinder array.	433
B.37	Modeled daily irradiance at MER.	434
B.38	Modeled noon irradiance at MER	435
B.39	Daily total solar energy—Spirit	435
B.40	Noon total solar energy—Spirit	436
B.41	Noon total solar energy—Opportunity	436
B.42	Daily total solar energy—Opportunity	437
B.43	Long-term power output of Opportunity.	438
B.44	Self-cleaning of Opportunity solar array	color
B.45	Effect of dust on Spirit solar arrays	438
B.46A	Solar cells coated with simulated Mars dust (“JSC 1”)	color
B.46B	Solar cells coated with simulated Mars dust (“Carbondale Clay”)	color
B.47	Obscuration of solar cells vs. dust loading.	441
C.1	High-resolution thermal inertia map of Mars.	color
C.2	Subsurface temperatures for a homogeneous subsurface	450
C.3	Subsurface temperatures for a two-layer subsurface	451
C.4	Geothermal gradient	452
C.5	Viking measurements of pressure on Mars.	453
C.5A	Global water vapor.	454
C.5B	Column abundance of water vapor on Mars	color
C.5C	Seasonally averaged water vapor column abundance on Mars.	color
C.5D	Water vapor abundance.	455
C.6	Subsurface ice—Leighton and Murray	458
C.7	Subsurface ice—Farmer and Doms.	459
C.8	Subsurface ice—Paige	460
C.9	Annual mean surface temperature.	461
C.10	Subsurface ice—Mellon and Jakosky.	462
C.11	Ice table depth	color
C.12	Subsurface ice—Schorghofer and Aharonson.	463
C.13	Atmospheric water vapor vs. obliquity	466
C.14	Stability of ground ice vs. Mars obliquity	470
C.15	Global variation of water content in upper ~1 m of Mars.	color
C.16	Depth and content of water layer—north	477
C.17	Depth and content of water layer—south	478
C.18	Maps showing distribution of eight properties on Mars	color
C.19	Albedo and thermal inertia on Mars	481
C.20	Distribution of water-equivalent hydrogen.	color

C.21	Water-equivalent hydrogen vs. elevation	482
C.22	Dependence of neutron flux on season	color
C.23	Phase diagram for water	487
C.24	Regions of Mars where $T > 273\text{ K}$	488
C.25	Illustration of polar sun angles	491
C.26	Gully profile	494
C.27	Model for underground aquifer	496
C.28	Dark streaks inside crater	497
C.29	Two images of the same area, with new streaks present in the later photo . .	497
C.30	Cryolithosphere.	499
C.31	Examples of the SLE and MLE ejecta morphologies	501
C.32	Ejecta morphology	503
C.33	Percentage of craters with ejecta due to volatiles	color
C.34	Percentage of craters that have DLE morphology	color
C.35	Distribution of onset diameters for rampart formation	color
C.36	Locations of DLE craters.	506
C.37	A possible distribution of H_2O in the Mars subsurface	color

Tables

2.1	State–Step table for mission to Mars orbit.	20
3.1	LSAM masses for ascent and descent systems	34
3.2	Effect of Δv on trip time	52
3.3	Mars “short-stay” missions	53
3.4	Cost per kg delivered to LEO.	55
3.5	JSC DRM-1 Launch Vehicle concepts.	56
3.6	Earth departure—fast and slow trips.	60
3.7	Mass sent toward Mars as % of mass in LEO.	61
3.8	Payload reduction to orbit vs. altitude	62
3.9	Fraction of mass in LEO sent toward Mars.	63
3.10	Mass of SEP orbit-raising system	64
3.11	Ascent systems from JSC DRMs	69
3.12	Fast and slow trips to Earth from Mars orbit	70
3.13	Gear ratios from LEO to Mars orbit—propulsion	77
3.14	Gear ratios from LEO to Mars orbit—aerocapture	78
3.15	Gear ratios from LEO to Mars surface—propulsion.	78
3.16	Gear ratios from LEO to Mars surface—aero-assist	78
3.17	IMLEO to Mars—propulsion.	79
3.18	IMLEO to Mars—aero-assist	79
3.19	Effect of NTP on Earth departure	80
3.20	Benefit factor from NTP	80
3.21	Mars orbit insertion using NTP	81
4.1	Consumable requirements	86
4.2	Consumable requirements for a Mars mission	89
4.3	Required IMLEO for consumables	90
4.4	Recommended organ dose limits.	93
4.5	LEO career dose limits	94
4.6	Risk in round-trip missions to Mars	99
4.7	DRM-1 Mars Transit/Surface Habitat	112
4.8	Comparison of Crew Lander masses DRM-1 vs. DRM-3	114

4.9	Comparison of ERV masses DRM-1 vs. DRM-3	115
4.10	Masses of Dual Lander Vehicles	117
4.11	Mass, power, and volume of MOB Habitat	119
4.12	MOB Habitat structural mass	120
4.13	MOB estimates for ECLSS	121
4.14	Sequence for direct entry	128
4.15	MSP 2001 Orbiter aerocapture system	131
4.16	Comparison of aerocapture and aerobraking	131
4.17	Mass breakdown for large entry vehicle	142
4.18	Brief summary of EDL roadmap—I	146
4.19	Brief summary of EDL roadmap—II	146
5.1	Key ISRU capabilities	157
5.2	FeO content of regolith	159
5.3	Power to extract oxygen from Mare regolith	159
5.4	Power to extract oxygen from highlands regolith	160
5.5	Power to drive off volatiles from regolith	162
5.6	Power to heat regolith to drive off water	164
5.7	Propellant mass/payload mass	178
5.8	IMLEO for orbit insertion, ascent, and departure—no ISRU	179
5.9	IMLEO for orbit insertion, ascent, and departure with ISRU	180
5.10	Unit mass and power requirements for Mars ISRU	189
5.11	Gross mass and power requirements for Mars ISRU	189
5.12	Δv for various lunar transfers	192
5.13	Sample spreadsheet for lunar water transfer	193
5.14	Sample spreadsheet for return from LL1 to Moon	195
5.15	Sample spreadsheet for transfer from LL1 to LEO	197
5.16	Sample spreadsheet for return from LEO to LL1	198
5.17	Mass of water transferred from lunar surface to LEO	199
5.18	Fraction of water transferred from lunar surface to LEO	199
5.19	Values of Δv for lunar transfers	199
5.20	Fraction of water transferred from Moon to LEO	200
5.21	Percentage of water transferred from Moon to lunar orbit	202
5.22	Payload sent toward Mars vs. level of staging	207
5.23	Δv for Mars long-stay round trip	207
5.24	Payload mass to Mars vs. level of staging—long stay	207
5.25	Δv for Mars short-stay round trip	208
5.26	Payload mass to Mars vs. level of staging—short stay	208
5.27	High-pressure hydrogen tanks	212
5.28	APL high-pressure hydrogen tank	213
5.29	Pressure–temperature–density data for hydrogen	213
5.30	Properties of liquid hydrogen	217
5.31	Hydrogen densities on metallic sorbents	221
5.32	% hydrogen by weight on metallic sorbents	221
6.1	Mass estimates for Earth Return Vehicle	262
6.2	Mass estimates for MAV/Cargo Lander	263
6.3	Mass estimates for Crew Lander	263
6.4	NASA estimates of masses for Mars vehicles	273
6.5	Reverse-engineered mass estimates for DAV and SH	273
6.6	Reverse-engineered mass estimates for CTV	274

6.7	Mass allocations for Mars Direct	275
6.8	Reverse engineering for Mars Direct ERV	276
6.9	Vehicles involved in the MSM	278
6.10	MSM vehicle mass budget	279
6.11	Mass budget for Habitats.	280
6.12	ISRU products in DRM-3 compared with MSM	280
6.13	ESA estimates for 2033 missions.	285
6.14	IMLEO for human mission to Mars	288
7.1	MIT-estimated values of IMLEO for three architectures and variable parameters	306
7.2	MIT-estimated values of payload delivered to the surface for three architectures and variable parameters	307
7.3	Lunar mission segment Δv and time of flight	312
7.4	Focused Study estimates of Δv requirements for lunar missions	313
7.5	Comparison of Focused Study and Broad Study estimates of Δv	315
7.6	Vehicle mass estimates	325
7.7	Steps in human lunar mission.	326
7.8	Example spreadsheet for Case 4	327
7.9	Summary of mass estimates for seven cases from Tables 7.8 and 7.9	328
8.1	JPL reverse-engineering of CTV	337
8.2	JPL reverse-engineering of Surface Habitat	337
8.3	JPL reverse-engineering of Descent/Ascent Vehicle	338
8.4	Comparison of NTR systems for TMI for three vehicles	338
8.5	2006 Exploration Systems Research and Technology (ESRT) Program.	352
B.1	Peak irradiance on a horizontal surface on Mars—no atmosphere	392
B.2	Daily total solar irradiance on Mars—no atmosphere.	392
B.3	Transmission coefficient vs. zenith angle and optical depth	400
B.4	Historical variation of Mars orbit.	422
B.5	Solar energy on tilted surfaces (northern hemisphere)	424
B.6	Solar energy on tilted surfaces (southern hemisphere)	425
C.1	Regions of ground ice stability within latitude range -60° to $+60^\circ$	468
C.2	Peak water content vs. elevation along six traverses	483
C.3	Excavated depths of craters	502

Abbreviations and acronyms

ALS	Advanced Life Support
AEDL	Aero Entry, Descent, and Landing (the process of using a aerodynamic forces to decelerate spacecraft at Mars or Earth for orbit insertion or descent)
AIAA	American Institute of Aeronautics and Astronautics
ALARA	As Low As Reasonably Achievable (refers to a philosophy for controlling radiation to astronauts)
ALS	Advanced Life Support (NASA program to develop advanced systems for life support)
Am	Amorphous
AM0	Area Mass
APL	Applied Physics Laboratory
ARC	NASA-Ames Research Center
AS	Ascent Stage (propulsion stage to ascend from Moon or Mars)
ASEE	American Society of Electrical Engineers
ASME	American Society of Mechanical Engineers
AU	Astronomical Unit
BAA	Broad Agency Announcement
BFO	Blood-Forming Organ (critical human element subjected to radiation).
BYA	Billion Years Ago
CaLV	Cargo Launch Vehicle (used to deliver ~125 mT of cargo to LEO)
CEV	Crew Exploration Vehicle (holds crew for transfer to lunar orbit—includes both CM and SM)
CI	Confidence Interval
CL	Cargo Lander

CLV	Crew Launch Vehicle (used to deliver crew in CEV to LEO)
CM	Crew Module (part of CEV that holds crew); Crew-Member
CRV	Crew Return Vehicle (a small crew capsule that tags along with the Habitat on the outbound trip to Mars to provide the crew with a back-up spacecraft)
CSI	Control-Structure Interaction (technical field dealing with control systems and structures)
CTV	Crew Transfer Vehicle (used to transfer crew from LEO to Mars and back)
DAV	Descent/Ascent Vehicle (combination of DS and AS)
DDT&E	Design, Development, Test, & Evaluation
DGB	Disk Gap Band
Di	Diverse
DIPS	Dynamic Isotope Power System
DLE	Double-Layer Ejecta
DOD	Department Of Defense
DOE	Department Of Energy
DPT	Decadal Planning Team (NASA committee)
DRM	Design Reference Mission (a paper analysis of requirements and potential systems for carrying out an end-to-end human mission to the Moon or Mars)
DS	Descent Stage (the propulsion stage used to descend from orbit to the surface)
ECLSS	Environmental Control & Life Support System (the system that controls the human environment in a Habitat and recycles resources)
EDL	Entry, Descent, and Landing (the process for orbit insertion and descent and landing at the Moon or Mars)
EDS	Earth Departure System (propulsion system for departure from LEO to go toward the Moon or Mars)
EHDP	bisphosphonate etidronate
EIRA	ESAS Initial Reference Architecture
ELV	Expendable Launch Vehicle
EOR	Earth Orbit Rendezvous (assembly point prior to departure for the Moon or Mars)
ERV	Earth Return Vehicle
ES	Exploration Systems (branch of NASA for human exploration)
ESA	European Space Agency
ESAS	Exploration Systems Architecture Study (2005 study of architecture for human return to Moon)
ESM	Equivalent System Mass
ESMD	Exploration Systems Mission Directorate (same as ES)
ESRT	Exploration Systems Research and Technology
ET	External Tank

ETSI	ExtraTerrestrial Solar Intensity
EVA	Extra-Vehicular Activity (operations by astronauts in suits outside of Habitats)
FOM	Figure Of Merit
GCR	Galactic Cosmic Ray (source of high-energy, low-level radiation in space)
GEO	Geostationary Earth Orbit
GRC	Glenn Research Center (NASA)
GSFC	Goddard Space Flight Center (NASA)
H	Habitat
HEO	High Earth Orbit (typically >10,000 km altitude)
HIMS	High Resolution Image Management System
HLLV	Heavy Lift Launch Vehicle
HLR	Human Lunar Return (a paper study by JSC in the mid-1990s)
HLV	Heavy Lift Vehicle
HQ	HeadQuarters
HZE	High charge and Energy for $Z > 3$
ICP	Intramural Call for Proposals
IMLEO	Initial Mass in Low Earth Orbit (total mass that must be transported to LEO from Earth to implement a space mission)
ISPP	<i>In Situ</i> Propellant Production (production of ascent propellants on the Moon or Mars from indigenous resources)
ISRU	<i>In Situ</i> Resource Utilization (production of useful products—e.g., ascent propellants—on the Moon or Mars from indigenous resources)
ISS	International Space Station
ITV	Interplanetary Transfer Vehicle (used to transfer crew from LEO to Mars and back)
JANNAF	Joint Army, Navy, NASA, Air Force Interagency Propulsion Committee
JIMO	Jupiter Icy Moon Orbiter (space mission to explore the icy moons of Jupiter using NEP; mission has not been funded)
JLMS	JPL/Lockheed-Martin Study in 2004–2005 (estimated masses and power needs of ISRU systems for Mars).
JPL	Jet Propulsion Laboratory (NASA)
JSC	Johnson Space Center (NASA)
L/D	Length/Diameter
L2	Libration Point
LaRC	Langley Research Center (NASA)
LCC	Life Cycle Cost (cost of a system over its life).
LCH ₄	Liquid methane

xxx **Abbreviations and acronyms**

LEO	Low Earth Orbit (typically, a circular orbit with altitude in the range 200 to 400 km)
LET	Linear Energy Transfer (process to extrapolate radiation impact data from one energy regime to another)
LH ₂	Liquid hydrogen
LANL	Los Alamos National Laboratory
LLO	Low Lunar Orbit (typically, a circular orbit of altitude 100 km)
LLT	LL1-to-LEO Tanker
LMO	Low Mars Orbit
LOC	Loss Of Crew (mission failure where crew dies)
LOI	Lunar Orbit Insertion (process to decelerate a spacecraft so as to inject it into lunar orbit)
LOM	Loss Of Mission (mission failure where crew is saved)
LOR	Lunar Orbit Rendezvous (process of transfer of crew from Ascent Vehicle to Earth Return Vehicle in lunar orbit)
LOX	Liquid OXYgen
LPS	Lunar Power System
LRO	Lunar Reconnaissance Orbiter (space mission to observe the Moon from orbit)
LS	Lunar Surface
LSAM	Lunar Surface Access Module (transports crew from lunar orbit to lunar surface and return)
LSH	Landing and Surface Habitat
LSR	Lunar Surface Rendezvous
LSS	Life Support System (provides food, water, air, and waste disposal to Habitats)
LST	Local Solar Time
LV	Launch Vehicle
LWT	Lunar Water Tanker
MAE	Materials Adherence Experiment
MARIE	Mars Radiation Environment Experiment
MARSIS	Mars Advanced Radar for Subsurface and Ionosphere Sounding
MAV	Mars Ascent Vehicle (vehicle used to transport crew from the surface to Mars orbit)
MEP	Mars Exploration Program
MEPAG	Mars Exploration Program Advisory Group
MER	Mars Exploration Rovers (robotic mission to Mars in 2004–2006)
MEx	Mars Express
MGS	Mars Global Surveyor (Mars orbiter)
MIT	Massachusetts Institute of Technology
MLE	Multiple-Layer Ejecta
MLI	Multi-Layer Insulation

MMH	Mono-Methyl Hydrazine (space-storable propellant)
MMOD	MicroMeteoroid/Orbital Debris
MMSSG	Moon–Mars Science Linkage Science Steering Group
MOB	Mars Or Bust (student project)
MOC	Mars Orbiter Camera
MOI	Mars Orbit Insertion (process to decelerate a spacecraft so as to inject it into Mars orbit)
MOLA	Mars Orbiter Laser Altimeter (measures altitude of surface on Mars)
MOR	Mars Orbit Rendezvous (process of transfer of crew from Ascent Vehicle to Earth Return Vehicle in Mars orbit)
MPLO	Mars PayLoad Orbit
MSFC	NASA-Marshall Space Flight Center
MSM	Mars Society Mission (DRM)
mT	metric Tons
MTV	Mars Transit Vehicle (serves as the interplanetary support vehicle for the crew for a round-trip mission to Mars orbit and back to Earth)
NASA	National Aeronautics and Space Administration
NASP	National AeroSpace Place
NCRP	National Council on Radiation Protection and Measurement
NEP	Nuclear Electric Propulsion (propulsion system using a nuclear reactor to generate electric power for electric propulsion).
NERVA	Nuclear Engine for Rocket Vehicle Application
NExT	NASA Exploration Team (NASA analysis team c. 2001)
NRC	National Research Council
NRPS	Nuclear Reactor Power System (used to supply power to outposts)
NS	Neutron Spectrometer (used to detect hydrogen in planetary surfaces)
NTO	Nitrogen TetrOxide (space-storable oxidant for rockets)
NTP	Nuclear Thermal Propulsion (same as NTR)
NTR	Nuclear Thermal Rocket (rocket that employs a nuclear reactor to heat hydrogen to a high temperature prior to efflux from a rocket nozzle)
OD	Optical Depth
OExP	Office of ExPLoration
OOS	On-Orbit Staging
ORU	Orbital Replacement Unit
OSHA	Occupational Safety and Health Administration
P(LOC)	Probability of Loss Of Crew
P(LOM)	Probability of Loss Of Mission
P/C	PhysicoChemical

PEM	Proton Exchange Membrane
PF	Mars PathFinder Mission
PMAD	Power Management And Distribution
Pn	Pancake
POD	Point Of Departure
pp	partial pressure
pr μm	precipitable microns
PSR	Precision Segmented Reflector (a now-defunct NASA program to develop a sub-millimeter telescope)
PVA	PhotoVoltaic Array
R&T	Research & Technology
RCS	Reaction Control System (propulsion system used for minor orbit corrections)
Rd	Radial delineated
REID	Risk of Exposure-Induced Death (effect of exposure of humans to radiation)
RFC	Regenerative proton exchange membrane Fuel Cell
RLEP	Robotic Lunar Exploration Program (NASA program to utilize robotic precursors to gain information prior to human landings on the Moon)
RQ	ReQuirements Division
RTG	Radioisotope Thermal Generator (device to convert heat from radioisotopes into electric power on spacecraft)
RTOP	Research/Technology Operation Plan
RWGS	Reverse Water-Gas Shift (chemical process to convert CO_2 to O_2)
S/E	Sabatier/Electrolysis
SAE	Society of Automotive Engineers
sccm	standard cubic centimeters per minute
SDHLV	Shuttle-Derived Heavy Lift Vehicle
SEI	Space Exploration Initiative (an abortive attempt by NASA to establish a human exploration initiative in the early 1990s)
SEOTV	Solar Electric Orbital Transfer Vehicle
SEP	Solar Electric Propulsion (propulsion system using solar arrays to generate electric power for electric propulsion)
SH	Surface Habitat
SHAB	Surface HABitat
SHARAD	SHAllow RADar on 2005 Mars Reconnaissance Orbiter
SI	<i>Système International</i> (international system of units of measurement)
SICSA	Sasakawa International Center for Space Architecture
SLE	Single-Layer Ejecta
SLH	SLush Hydrogen

SM	Service Module (part of CEV that supplies resources to CM and propels CM back to Earth)
SOE	Solid-Oxide Electrolysis
SPE	Solar Particle Event (sudden process in which large flux of protons is emitted by the Sun that travel through the solar system)
SPS	Solar Power Satellite
SRM	Solid Rocket Motor (used to boost capability of Launch Vehicles)
SS	Space Shuttle
SSE	Space Science Enterprise
SSME	Space Shuttle Main Engine
STS	Space Transportation System
TAO	Take-off and Ascent to Orbit (TAO)
TEI	Trans-Earth Injection (the process of using propulsion to depart from Mars orbit or lunar orbit and head toward Earth)
TES	Thermal Emission Spectrometer
THEMIS	Thermal Emission Imaging System
TLI	Trans-Lunar Injection (the process of using propulsion to depart from LEO and head toward the Moon)
TMI	Trans-Mars Injection (the process of using propulsion to depart from LEO and head toward Mars)
TPH	Triple-Point hydrogen
TPS	Thermal Protection System (protects spacecraft from extreme heat generated in aero-entry)
TRL	Technology Readiness Level
TSH	Transfer and Surface Habitat
UVVIS	UltraViolet and VISible
W&R	Whedon & Rambaut (authors of paper on effects of exposure to zero- <i>g</i>)
WEH	Water-Equivalent Hydrogen
WEU	Water Extraction Unit (conceptual system to remove water from putative ice-containing regolith on Moon)
YBP	Years Before Present
YSZ	Yttria-Stabilized Zirconia (used for solid-state electrolysis of CO ₂)
ZBO	Zero Boil-Off